

Amendments to the Claims:

This listing of claims will replace all prior versions, and listings, of claims in the application:

Listing of Claims:

1-11. (Cancelled)

12. (New) A method of forming a coating on a substrate using a low pressure plasma spray, using a coating material in the form of a powder beam for spraying onto a surface of a substrate comprising:

operating the plasma spray to produce a plasma stream which delivers the coating material to the substrate, wherein the coating material in the powder beam is at least partially melted, the operating including introducing a plasma gas into a plasma gun to establish plasma gas operating conditions;

generating a plasma intensity sufficiently high enough to vaporize approximately 5-30% of the powder coating material;

maintaining a powder conveying rate sufficiently low enough to form an anisotropically structured coating having anisotropic columnar microstructures aligned perpendicular to the substrate surface having transitional zones in which material-deficient zones bound the columnar particles at their sides;

maintaining a process pressure sufficiently low enough to form an anisotropically structured coating having anisotropic columnar microstructures aligned perpendicular to the substrate surface having transitional zones in which material-deficient zones bound the columnar particles at their sides; and

maintaining a gas flow rate sufficiently low enough to form an anisotropically structured coating having anisotropic columnar microstructures aligned perpendicular to the

20 substrate surface having transitional zones in which material-deficient zones bound the columnar
21 particles at their sides.

1 13. (New) The method of claim 12 wherein the process pressure is lower than
2 10,000 Pa.

1 14. (New) The method of claim 12 wherein the anisotropically structured
2 coating comprises a heat insulating coating having a coating thickness between 20 and 1000 μm ,
3 said heat insulating coating being built up from a plurality of layers.

1 15. (New) The method of claim 12 wherein the anisotropically structured
2 coating comprises a heat insulating coating having a coating thickness greater than 100 μm , said
3 heat insulating coating being built up from a plurality of layers.

1 16. (New) The method of claim 12 wherein the process pressure is in the
2 range between 50 and 2000 Pa.

1 17. (New) The method of claim 12 wherein the process pressure is in the
2 range between 100 and 800 Pa.

1 18. (New) The method of claim 12 wherein the specific enthalpy of the
2 plasma is produced by emitting an effective power that is in the range from 40 to 80 kW.

1 19. (New) The method of claim 12 wherein the powder beam is injected into
2 the plasma with a process gas comprising a mixture of inert gases having a total gas flow in the
3 range from 30 to 150 SLPM.

1 20. (New) The method of claim 19 wherein the mixture of inert gases
2 comprises a mixture of argon, Ar, and helium, He, with the volume ratio of Ar to He being in the
3 range from 2 : 1 to 1 : 4.

1 21. (New) The method of claim 12 wherein the powder delivery rate is
2 between 5 and 60 g/min.

1 22. (New) The method of claim 12 wherein the powder delivery rate is
2 between 10 and 40 g/min.

1 23. (New) The method of claim 12 wherein the substrate is moved with
2 rotational movements relative to a cloud of defocused powder beam during the coating.

1 24. (New) The method of claim 12 wherein the substrate is moved with
2 pivotal movements relative to a cloud of defocused powder beam during the coating.

1 25. (New) The method of claim 12 wherein the coating material comprises
2 oxide ceramic components.

1 26. (New) The method of claim 25 wherein the oxide ceramic components
2 comprises a material selected from the group consisting of zirconium oxide fully stabilized with
3 yttrium, cerium or other rare earths and zirconium oxide partly stabilized with yttrium, cerium or
4 other rare earths,
5 and wherein the material used as the stabilizer is added as an alloy to the
6 zirconium oxide in the form of an oxide of the said rare earths.

1 27. (New) The method of claim 12 wherein the coating material has a particle
2 size distribution as determined by a laser scanning method in the range between 1 and 50 μm .

1 28. (New) The method of claim 12 wherein the coating material has a particle
2 size distribution as determined by a laser scanning method in the range between 3 and 25 μm .

 29. (New) The method of claim 12 wherein the coating material's powdery
particles are manufactured by a spray drying or a combination of melting and subsequent
breaking and/or grinding.

1 30. (New) The method of claim 12 further comprising
2 using an additional heat source configured for forming the coating on the
3 substrate within a predetermined temperature range.

1 31. (New) The method of claim 30 wherein the predetermined temperature
2 range is between 300 and 900 °C.

1 32. (New) The method of claim 30 wherein the predetermined temperature
2 range is between 450 and 750 °C.

1 33. (New) The method of claim 12 wherein said anisotropically structured
2 coating comprises a heat insulating layer.

1 34. (New) The method of claim 12 wherein said coating on said substrate
2 comprises a heat insulating coating system, said heat insulating coating system including a heat
3 insulating coating, and a base coating between the substrate and the heat insulating coating,
4 wherein said heat insulating coating system is applied in a single work cycle by low pressure
5 plasma spray (LPPS) thin film processes.

1 35. (New) The method of claim 34 wherein said substrate is a substrate
2 selected from the group consisting of a Ni base alloy, and a Co base alloy.

1 36. (New) The method of claim 34 wherein said substrate is part of a
2 component selected from the group consisting of a component of a stationary gas turbine, a
3 component of an airplane power plant turbine vane, a component of an airplane power plant
4 guide vane, a component of an airplane power plant turbine blade, a component which can be
5 exposed to a hot gas, and a heat shield.

1 37. (New) The method of claim 34 further comprising heat treating said
2 substrate comprising said heat insulating coating system.

1 38. (New) The method of claim 34 wherein said base coating includes a hot
2 gas corrosion protection coating, whose coating thickness has a value between 10 and 300 μm ,
3 and which comprises at least in part, a metal aluminide of an MeCrAlY alloy, with Me standing
4 for one of the metals Fe, Co or Ni, or a ceramic oxide material which preferably has a structure
5 selected from the group consisting of a dense, a columnar, a directional, and a unidirectional
6 structure.

1 39. (New) The method of claim 34 wherein said base coating includes a hot
2 gas corrosion protection coating, whose coating thickness has a value between 25 and 150 μm ,
3 and which comprises at least in part, a metal aluminide of an MeCrAlY alloy, with Me standing
4 for one of the metals Fe, Co or Ni, or a ceramic oxide material which preferably has a structure
5 selected from the group consisting of a dense, a columnar, a directional, and a unidirectional
6 structure.

1 40. (New) The method of claim 34 wherein said coating on said substrate
2 further comprises a cover coating on the heat insulating coating.

1 41. (New) The method of claim 40 wherein said top coating comprises a
2 smoothing coating whose coating thickness has a value between 1 and 50 μm and which
3 comprises at least in part the same material as or a similar material to the heat insulating coating.

1 42. (New) The method of claim 40 wherein said top coating comprises a
2 smoothing coating whose coating thickness has a value between 10 and 30 μm and which
3 comprises at least in part the same material as or a similar material to the heat insulating coating.